ANALYSIS ON EFFECT OF EXCITATION FREQUENCY ON NONLINEAR CHARACTERISTICS OF A GEAR PAIR SYSTEM BASED ON NUMERICAL SIMULATION

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ABSTRACT

A nonlinear dynamic model of the geared rotor system, with consideration of the bearing clearances and gear backlashes, is established. A numerical algorithm based on Runge–Kutta-Butcher method is used to study the nonlinear behaviors of the system accounting for the effects of different excitation frequency. A detailed parametric research is conducted to analyze the influences of the excitation frequency on the nonlinear characteristics of the system. The final goal of the work is to study bifurcation and chaotic behavior of a gear pair model that includes nonlinearities due to bearing and teeth backlash. The analyses are performed in a parametric manner using non-dimensional excitation as bifurcation parameters. The results show that the different excitation frequency can significantly influence the nonlinear dynamic responses.

Keywords: *Nonlinear Response, Excitation Frequency, Bifurcation Phenomena*

1. INTRODUCTION

Geared rotor systems can be found in a wide range of automotive, aerospace, marine, wind turbine and industrial applications. The dynamic response of a geared rotor system, in generally, depends on the time-varying stiffness, static transmission error, gear backlash, the damping coefficient ratio and excitation loads. These parameters have direct effects on the dynamic response and performance of the system. The nonlinear dynamic behaviors of the system are investigated and the several studies on the effect of systematic parameter on the dynamic response are summarized as follows.

A considerable amount of studies on the dynamic model for the dynamic analysis of geared rotor system are reported by many researchers[1,2,3,4]. Raghothama and Narayanan [5] carried out the incremental harmonic balance method to investigate the periodic motions of a non-linear geared rotor-bearing system. It was conducted that the system exhibited a period doubling route and a quasi-periodic route to chaos in different regions of excitation frequency. Vedmar and Anersson[6] presented a method to calculate dynamic gear tooth force and bearing forces where the bearing model was under elastic bearings assumption. Alhyyab and Kahraman[7] investigated steady state period-one motions of a nonlinear geared rotor-bearing

system by using a multi-term harmonic balance method in conjunction with discrete Fourier transforms. Rui and Chen[8] analyzed the dynamic response of gear model under primary resonance and 1/2 sub-harmonious resonance by means of averaging method. Vaishya and Singh[9] used a sliding friction method to simulate the nonlinear dynamics of a gear system in order to obtain insights into the relative effects of sliding friction and mesh damping, respectively. Liu and Robert[10] examined a nonlinear, parametrically excited dynamics of idler gear sets. The periodic steady state solutions were obtained using analytical and numerical approaches. Li et al[11] applied the theory of multiple scales method to research the vibration behaviors of gear system.

In this study, The paper proposes a study of the nonlinear characteristics of a gear pair. A numerical algorithm based on Runge–Kutta-Butcher (RKB) method is used to study the nonlinear behaviors of a gear pair accounting for the effects of the excitation frequency. A detailed parametric research is conducted to analyze the influences of the excitation frequency on the nonlinear characteristics of the system. Nonlinear dynamic responses accompanied by nonlinear phenomena like bifurcation, periodic and quasi-periodic solutions and chaos are investigated. Different routes to chaos for the geared rotor system are illustrated and the evolutions of the dynamic responses are demonstrated by phase plane plots, Poincaré Maps,

fast Fourier transform spectra and bifurcation diagrams in detail.

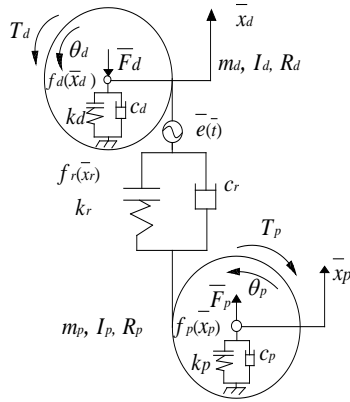## 2. MODELING AND EQUATION OF MOTION

Figure 1: Nonlinear Model Of A Gear Pair System

The nonlinear model of a geared rotor system considered in this study is shown in Figure1. In this model it is assumed that the system is symmetric about the plane of the gears and the axial motion is negligible. Bearings and shafts that support the gears are represented by equivalent damping, linear stiffness and clearance elements as shown in Figure1. The damping elements are characterized by linear viscous damping coefficients c_d and c_p The bearing clearance elements are defined by force-displacement function $f_d(\overline{x_d})$ and $f_p(\overline{x_p})$. The support stiffness elements are represented by corresponding scaling constants k_d and k_p External radial preloads are applied to both the rolling element bearings. Therefore, the essential vibration of the system can be described by four degrees of freedom system, with coordinates

$$X = [\theta_d \quad \theta_p \quad \overline{x}_d \quad \overline{x}_p]^T \qquad (1)$$

By using Newton's second law, the corresponding equations of vibration are obtained in the general form

$$\begin{cases} I_d \ddot{\theta}_d + R_d h_r(\theta_d, \overline{x}_r, \dot{\overline{x}}_r) = T_d(\overline{t}, \theta_d, \dot{\theta}_d) \\ I_p \ddot{\theta}_p - R_p h_r(\theta_d, \overline{x}_r, \dot{\overline{x}}_r) = -T_p(\overline{t}, \theta_p, \dot{\theta}_p) \\ m_d \ddot{\overline{x}}_d + h_d(\overline{x}_d, \dot{\overline{x}}_d) + h_r(\theta_d, \overline{x}_r, \dot{\overline{x}}_r) = \overline{F}_d \\ m_d \ddot{\overline{x}}_d + h_d(\overline{x}_d, \dot{\overline{x}}_d) + h_r(\theta_d, \overline{x}_r, \dot{\overline{x}}_r) = \overline{F}_d \\ m_p \ddot{\overline{x}}_p + h_p(\overline{x}_p, \dot{\overline{x}}_p) - h_r(\theta_d, \overline{x}_r, \dot{\overline{x}}_r) = \overline{F}_p \end{cases} \qquad (2)$$

This model takes into account the static transmission, $\overline{e}(\overline{t})$ representing geometrical errors

of the gear teeth profile and spacing. In addition, the quantity

$$\overline{x}_r(\overline{t}) = R_d \theta_d - R_p \theta_p + \overline{x}_d(\overline{t}) - \overline{x}_p(\overline{t}) - \overline{e}(\overline{t}) \qquad (3)$$

is known as the dynamic transmission error.

The damping mechanism in both the gear mesh and the bearings are assumed to be linear, so that the corresponding forces may be expressed in the simpler form

$$\begin{cases} h_r(\theta_d, \overline{x}_r, \dot{\overline{x}}_r) = c_r \dot{\overline{x}}_r + h_r(\theta_d, \overline{x}_r) \\ h_n(\overline{x}_n, \dot{\overline{x}}_n) = c_n \dot{\overline{x}}_n + f_n(\overline{x}_n) \quad n = (d, p) \end{cases} \qquad (4)$$

In these equations, $\overline{x}_d$ and $\overline{x}_p$ are the transverse displacements of the gears; θ_d and θ_p are the wheel angular positions; m_d and m_p are the gear masses; I_d and I_p are the mass moments of inertia of the gears; R_d and R_p are the base circle radii of the gears. k_r is the gear mesh stiffness; c_r represents the linear viscous damping of the gear mesh; the quantity $T_d(\overline{t}, \theta_d, \dot{\theta}_d)$ represents the moment input to the driving gear, while, $T_p(\overline{t}, \theta_p, \dot{\theta}_p)$ is the resistant moment developed on the driven gear during the motion. $\overline{F}_d$ and $\overline{F}_p$ refer to the bearing pretension radial forces applied on the gear centers.

Furthermore, taking into account the gear backlash, the gear meshing restoring force component is expressed in the form

$$h_r(\theta_d, \overline{x}_r) = f_r(\overline{x}_r) k_r \qquad (5)$$

where k_r represents the gear mesh stiffness, while

$$f_r(\overline{x}_r) = \begin{cases} \overline{x}_r - l & \overline{x}_r \geq l \\ 0 & |\overline{x}_r| \leq l \\ \overline{x}_r + l & \overline{x}_r \leq -l \end{cases} \qquad (6)$$

and $2l$ represents the total gear backlash. Likewise, the restoring forces in the bearings is expressed in the classical linear form

$$f_n(\overline{x}_n) = k_n \overline{x}_n \qquad (7)$$

Next, introducing the parameters

$$\begin{cases} \bar{x}_R = R_d \theta_d - R_p \theta_p \\ m_e = I_d I_p \big/ R_d{}^2 I_p + R_p{}^2 I_d \end{cases} \quad (8)$$

The original set of equations of motion (2) can be obtained in the following form

$$\begin{cases} m_d \ddot{\bar{x}}_d + c_d \dot{\bar{x}}_d + c_r(\dot{\bar{x}}_R + \dot{\bar{x}}_d - \dot{\bar{x}}_p - \dot{\bar{e}}(\bar{t})) \\ \quad + k_d f_d(\bar{x}_d) + k_r f_r(\bar{x}_R + \bar{x}_d - \bar{x}_p - \bar{e}(\bar{t})) = -\bar{F}_d \\ m_p \ddot{\bar{x}}_p + c_p \dot{\bar{x}}_p - c_r(\dot{\bar{x}}_R + \dot{\bar{x}}_d - \dot{\bar{x}}_p - \dot{\bar{e}}(\bar{t})) \\ \quad + k_p f_p(\bar{x}_p) - k_r f_r(\bar{x}_R + \bar{x}_d - \bar{x}_p - \bar{e}(\bar{t})) = \bar{F}_p \\ m_e \ddot{\bar{x}}_R + c_r(\dot{\bar{x}}_R + \dot{\bar{x}}_d - \dot{\bar{x}}_p - \dot{\bar{e}}(\bar{t})) \\ \quad + k_r f_r(\bar{x}_R + \bar{x}_d - \bar{x}_p - \bar{e}(\bar{t})) = \bar{F}_{Ro} \end{cases} \quad (9)$$

Dimensionless forms of the above equations are obtained by letting

$$\begin{cases} x_i(t) = \bar{x}_i(\bar{t}) \big/ b, \, x_r(t) = \bar{x}_r(\bar{t}) \big/ b, \, \omega_n = \sqrt{k_r / m_e} \\ \omega_i = \sqrt{k_i / m_i} \, (\text{i} = \text{p,d}), t = \omega_n \bar{t}, \omega_i = \bar{\omega}_i / \omega_n \\ k_{13} = m_e / m_d, \zeta_{13} = c_r / (2 m_d \omega_n), F_i = \bar{e} / (b) \\ F_d = \bar{F}_d \big/ (m_d b \omega_n{}^2), F_p = \bar{F}_p \big/ (m_p b \omega_n{}^2) \\ k_{23} = m_e / m_p, F_{Ro} = \bar{F}_{Ro} \big/ (m_e b \omega_n{}^2) \\ F_o = \bar{F}_o \big/ (m_e b \omega_n{}^2), \bar{F}_{Ro} = T_d / R_d \end{cases} \quad (10)$$

The dimensionless governing equations of motion (9) are yielded in the normalized form

$$M\ddot{U} + C\dot{U} + Kf = F(t) \quad (11)$$

with

$$f_i(x_i) = \begin{cases} x_i - b_i / b & x_i > b_i / b \\ 0 & -b_i / b \le x_i \le b_i / b \quad i = d, p, r \\ x_i + b_i / b & x_i < -b_i / b \end{cases} \quad (12)$$

$$F(t) = \begin{pmatrix} -F_d \\ F_p \\ F_{Ro} \end{pmatrix} - \begin{pmatrix} 0 \\ 0 \\ F_i \omega_i{}^2 \end{pmatrix} \sin(\omega_i t + \psi_i) + \begin{pmatrix} 0 \\ 0 \\ F_o \end{pmatrix} \sin(\omega_o t + \psi_o) \quad (13)$$

$$\begin{cases} M = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ -1 & 0 & 1 \end{bmatrix} \\ C = 2 \begin{bmatrix} \zeta_{11} & 0 & \zeta_{13} \\ 0 & \zeta_{22} & -\zeta_{23} \\ 0 & 0 & \zeta_{33} \end{bmatrix} \\ K = \begin{bmatrix} k_{11} & 0 & k_{13} \\ 0 & k_{22} & -k_{23} \\ 0 & 0 & 1 \end{bmatrix} \\ f = \begin{pmatrix} f_d(x_d) \\ f_p(x_p) \\ f_r(x_r) \end{pmatrix} \end{cases} \quad (14)$$

3. NUMERICAL SOLUTIONS

The nonlinear dynamic equations presented in equation (11) for the geared rotor system are solved by using the fifth order Runge–Kutta-Butcher method. The algorithm is written as

$$y_{i+1} = y_i + \frac{1}{90}(7k_1 + 32k_3 + 12k_4 + 32k_5 + 7k_6)h \quad (15)$$

where

$$\begin{cases} k_1 = f(t_i, y_i) \\ k_2 = f(t_i + \frac{1}{4}h, y_i + \frac{1}{4}k_1 h) \\ k_3 = f(t_i + \frac{1}{4}h, y_i + \frac{1}{8}k_1 h + \frac{1}{8}k_2 h) \\ k_4 = f(t_i + \frac{1}{2}h, y_i - \frac{1}{2}k_2 h + k_3 h) \\ k_5 = f(t_i + \frac{3}{4}h, y_i + \frac{3}{16}k_1 h + \frac{9}{16}k_4 h) \\ k_6 = f(t_i + h, y_i - \frac{3}{7}k_1 h + \frac{2}{7}k_2 h + \frac{12}{7}k_3 h - \frac{12}{7}k_4 h + \frac{8}{7}k_5 h) \end{cases} \quad (16)$$

Note that h is the time step; t_i and y_i are the current values and $f(t_i, y_i)$ is called an increment function which can be interpreted as representative slope over the interval. The initial value of the h in the iterative solution procedure is assigned a value of $\pi/500$ and the termination criterion is specified as an error tolerance of less than 0.00002. The time series data corresponding to the first 1000 revolutions of the two gears are deliberately excluded from the dynamic analysis to ensure that the analyzed data relate to steady-state conditions. The sampled data are used to generate the Phase diagrams, the Poincaré maps, the power spectra and the bifurcation diagrams of the geared rotor system for obtaining a basic understanding of dynamic behavior.
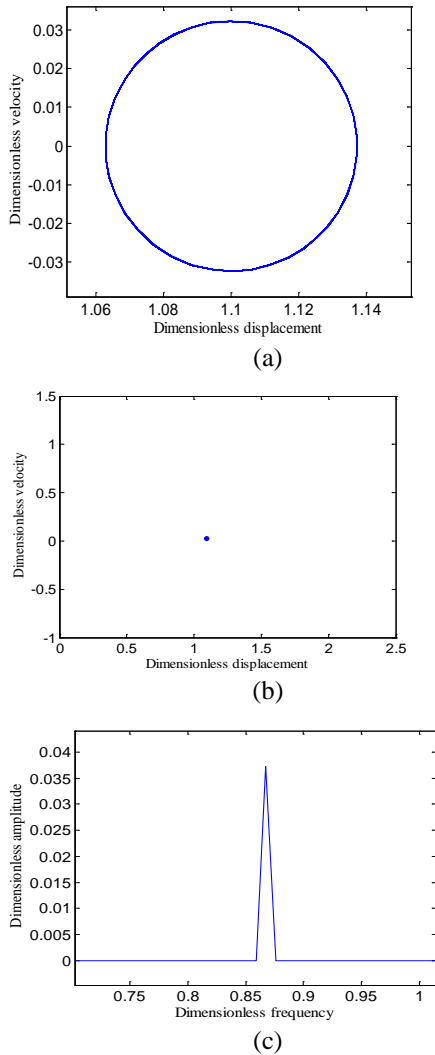
4. RESULTS AND DISCUSSIONS

(a)

(b)

(c)

Figure 2: Simulation Obtained For k_{11} =0.25, k_{22} =0.25 With ω_i =0.8675,

(A) Phase Diagram; (B) Poincaré Map; (C) Power Spectrum.

To test the existence of the essential dynamics of the geared rotor system and provide valuable insights into its nonlinear dynamic behavior, the phase diagrams, the Poincaré maps and the power spectra of the system are illustrated in Figures.2-5.

Figure 2 shows simulation obtained for k_{11} =0.25, k_{22} =0.25 with ω_i =0.8675. From Figure2, we can see that the spectrum graph includes mesh frequency 0.8675; the phase plane is a circle; the Poincaré map is a point. It shows that the vibration

of the system is a simple harmonic motion.

Figure 3 presents simulation obtained for k_{11} =0.25, k_{22} =0.25 with ω_i =1.1256. From Figure 3, we can observe that the spectrum graph includes mesh frequency 1.1256 and n1.1256/3 (n=0, ±1, ±2…) frequencies; the phase plane is a 3-looped circle; the Poincaré map includes three points. It is concluded that the vibration of the system is 1/3 sub-harmonic motion.
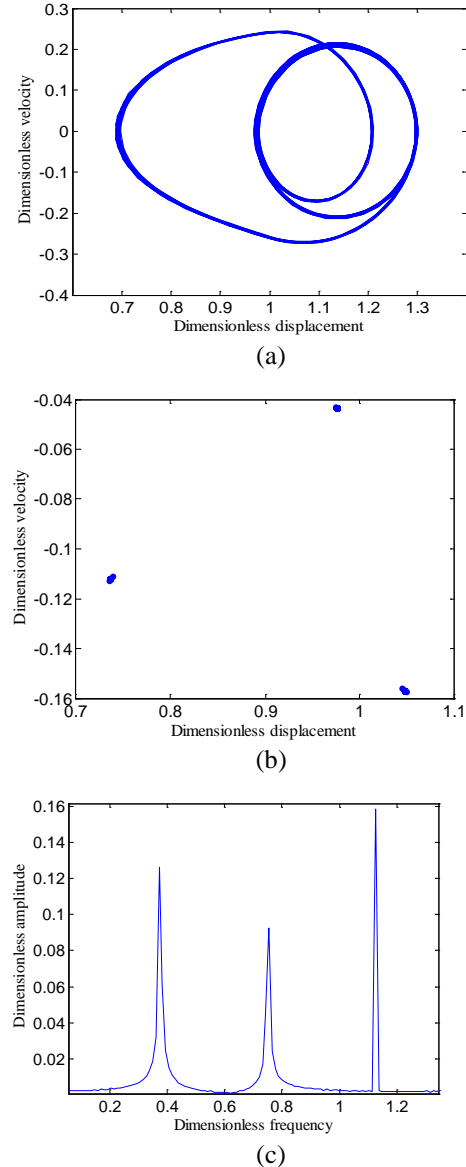

(a)

(b)

(c)

Figure 3: Simulation Obtained For k_{11} =0.25, k_{22} =0.25 With ω_i =1.1256. (A) Phase Diagram; (B) Poincaré Map; (C) Power Spectrum.

Figure4 depicts simulation obtained for k_{11}=0.25, k_{22}=0.25 with ω_i =1.4060. From Figure 4, it can be

found that the spectrum graph consists of two fundamental frequencies ω_1 and ω_2 at a ratio ω_1/ω_2 and there are peaks at the combination frequencies $m\omega_1+n\omega_2$ (m, $n=0$, ±1, $\pm2\ldots$); the phase plane is a shaped belt; the Poincaré map is a deformed closed orbit. It is implied that the vibration of the system is quasi-periodic motion.
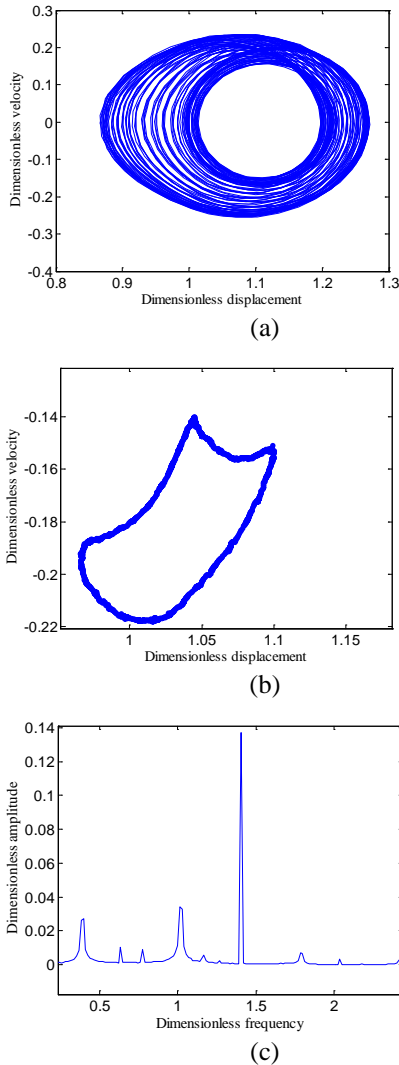

(a)

(b)

(c)

Figure 4: Simulation Obtained For k_{11} =0.25, k_{22} =0.25 With ω_i =1.4060. (A) Phase Diagram; (B) Poincaré Map; (C) Power Spectrum.

Figure 5 denotes simulation obtained for k_{11} =0.25, k_{22} =0.25 with ω_i =1.4060. From Figure 5, we can see that a characteristic broad spectrum is obtained; the phase plane is a chaotic strange attractor; the Poincaré map includes many points. It is concluded that the vibration of the system is chaotic motion.
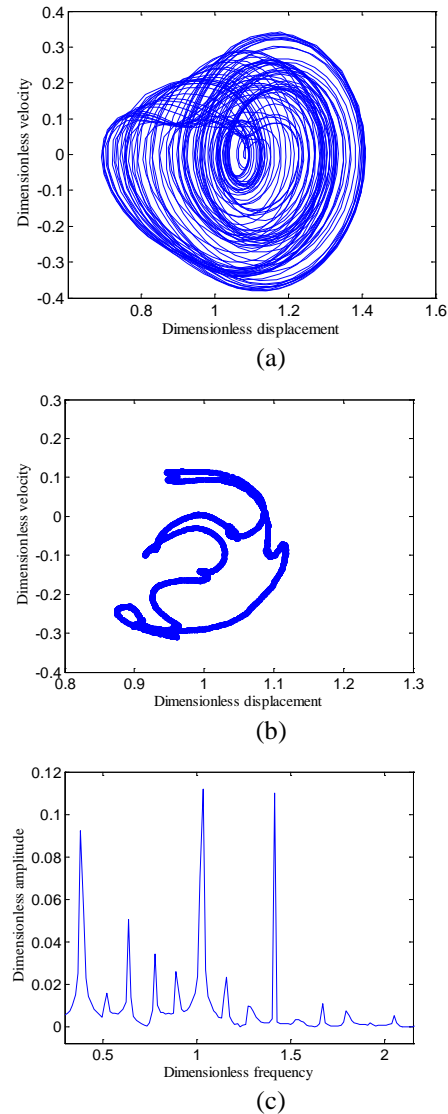

(a)

(b)

(c)

Figure 5: Simulation Obtained For k_{11} =0.25, k_{22} =0.25 With ω_i =1.4150. (A) Phase Diagram; (B) Poincaré Map; (C) Power Spectrum.

Figure 6(a) presents bifurcation diagram for the meshing gears in the line of action direction for k_{11} =0.25, k_{22} =0.25. With the changes of excitation frequency ω_i the vibration of the system also changes. It can be observed that geared rotor system exhibits periodic motion at almost of the dimensionless coefficient of ω_i, i.e. $\omega_i < 1.1256$. Furthermore, when ω_i is 1.0, the mesh frequency closes to the natural frequency of the system and the jumping phenomenon occurs. However, at a higher dimensionless ration of ω_i = 1.4060, the system undergoes quasi-periodic motion. It is

characterized by chaotic motion in the intervals with ω_i = 1.2335-1.4057 and ω_i > 1.4062. The results show that the system vibrations change from periodic motion to chaotic motions through 3-period bifurcation, and changes from chaotic motions to periodic motion by inverse quasi-period bifurcation.
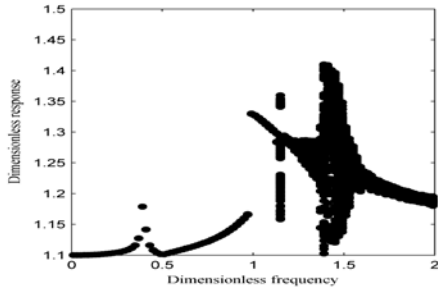
(a)
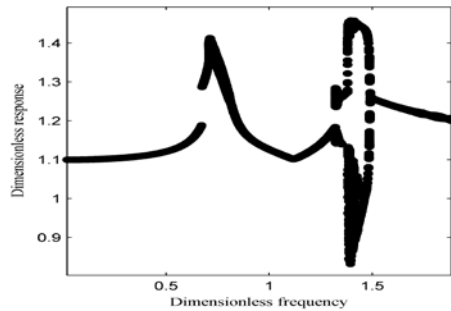
(b)

Figure 6: Bifurcation Diagrams Of Meshing Gears In The Contact Line Direction Using Dimensionless Frequency, ω_i As Bifurcation Parameter With

$F_i =0.0532,\ \xi_{33} =0.0516:$ *(A)* $k_{11} =0.25,\ k_{22} =0.25;$

(B) $k_{11} =1.75,\ k_{22} =1.75.$

Figure 6(b) shows bifurcation diagram for the meshing gears in the line of action direction for $k_{11} =1.75$, $k_{22} =1.75$. Comparison of the bifurcation diagram with those presented in Figure 6(a), it can be seen that the system shows 1T-periodic motion at low values of the dimensionless frequency, i.e. $\omega_i \leq 1.3253$. However, as ω_i is increased from 1.3253 to 1.3673, the 1T-periodic motion is replaced by sub-synchronous 2T-periodic motion. Chaotic behavior appears as the dimensionless frequency is further increased from ω_i =1.3673 to ω_i =1.4923. The system also exhibits periodic motion at high values of the dimensionless frequency, i.e. ω_i > 1.4923. Moreover, there are jumping phenomena at the dimensionless frequency

of ω_i =0.6051 and ω_i =1.3675. Thus, it can be inferred that the dynamic responses change from periodic motions to chaotic motions by double-period bifurcation and from chaotic motions to periodic motions by inverse double-period bifurcation.

5. CONCLUSIONS

In the present paper, the nonlinear model is proposed to study the dynamic behaviors of the gear pair based on the numerical method of the RKB. The results show that the different excitation frequency can significantly influence the nonlinear dynamic response. There are three kinds of forced stead responses, i.e. simple harmonic, sub-harmonic and chaotic responses. The system responses transform from periodic motions to chaotic motion through double-period bifurcation and from chaotic motions to periodic motions by inverse double-period bifurcation.

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